

$$\textcircled{1} \quad (a) \quad \rho = \frac{\text{mass}}{\text{area}} = \frac{m}{L_1 L_2}.$$

$$\begin{aligned} (b) \quad I_1^{R/O} &= \rho \int p \times (n_1 \times p) dA \\ &= \rho \int_0^{L_2} \int_0^{L_1} (x_2^2 n_1 - x_1 x_2 n_2) dx_1 dx_2 \\ &= \frac{m}{L_1 L_2} \left(n_1 \frac{L_1 L_2^3}{3} - n_2 \frac{L_1^2 L_2^2}{4} \right) \\ &= mL_2 \left(\frac{L_2}{3} n_1 - \frac{L_1}{4} n_2 \right) \end{aligned}$$

$$p = x_1 n_1 + x_2 n_2$$

$$n_1 \times p = x_2 n_3$$

$$n_2 \times p = -x_1 n_3$$

$$n_3 \times p = -x_2 n_1 + x_1 n_2$$

$$r = -y_1 n_1 - y_2 n_2$$

$$n_1 \times r = -y_2 n_3$$

$$n_2 \times r = y_1 n_3$$

$$n_3 \times r = y_2 n_1 - y_1 n_2$$

$$\begin{aligned} (c) \quad I_3^{R/O} &= \rho \int r \times (n_3 \times r) dA \\ &= \rho \int_0^{L_2} \int_0^{L_1} (y_1^2 + y_2^2) n_3 dy_1 dy_2 = \frac{m}{L_1 L_2} \left(\frac{L_1^3 L_2}{3} + \frac{L_1 L_2^3}{3} \right) n_3 \\ &= \frac{m}{3} (L_1^2 + L_2^2) n_3 \end{aligned}$$

$$\begin{aligned} (d) \quad I_2^{R/O} &= \frac{m}{L_1 L_2} \int_0^{L_2} \int_0^{L_1} p \times (n_2 \times p) dx_1 dx_2 = \frac{m}{L_1 L_2} \int_0^{L_2} \int_0^{L_1} (x_1^2 n_2 - x_1 x_2 n_1) dx_1 dx_2 \\ &= mL_1 \left(-\frac{L_2}{4} n_1 + \frac{L_1}{3} n_2 \right) \end{aligned}$$

$$\begin{aligned}
 I_3^{R/O} &= \frac{m}{L_1 L_2} \int_0^{L_2} \int_0^{L_1} \rho \times (\mathbf{n}_3 \times \rho) dx_1 dx_2 = \frac{m}{L_1 L_2} \int_0^{L_2} \int_0^{L_1} (x_1^2 + x_2^2) n_3 dx_1 dx_2 \\
 &= \frac{m}{3} (L_1^2 + L_2^2) n_3.
 \end{aligned}$$

Hence, we have

$$I^{R/O} = \begin{bmatrix} \frac{m}{3} L_2^2 & -\frac{m}{4} L_1 L_2 & 0 \\ -\frac{m}{4} L_1 L_2 & \frac{m}{3} L_1^2 & 0 \\ 0 & 0 & \frac{m}{3} (L_1^2 + L_2^2) \end{bmatrix}.$$

Let $\mathbf{r} = y_1 \mathbf{n}_1 + y_2 \mathbf{n}_2$ be the vector from R^* to an arbitrary point.

$$\begin{aligned}
 I_1^{R/R^*} &= \frac{m}{L_1 L_2} \int_{-\frac{L_2}{2}}^{\frac{L_2}{2}} \int_{-\frac{L_1}{2}}^{\frac{L_1}{2}} \mathbf{r} \times (\mathbf{n}_1 \times \mathbf{r}) dy_1 dy_2 = \frac{m}{L_1 L_2} \int_{-\frac{L_2}{2}}^{\frac{L_2}{2}} \int_{-\frac{L_1}{2}}^{\frac{L_1}{2}} (y_2^2 \mathbf{n}_1 - y_1 y_2 \mathbf{n}_2) dy_1 dy_2 \\
 &= \frac{m}{12} L_2^2 \mathbf{n}_1
 \end{aligned}$$

$$\begin{aligned}
 I_2^{R/R^*} &= \frac{m}{L_1 L_2} \int_{-\frac{L_2}{2}}^{\frac{L_2}{2}} \int_{-\frac{L_1}{2}}^{\frac{L_1}{2}} \mathbf{r} \times (\mathbf{n}_2 \times \mathbf{r}) dy_1 dy_2 = \frac{m}{L_1 L_2} \int_{-\frac{L_2}{2}}^{\frac{L_2}{2}} \int_{-\frac{L_1}{2}}^{\frac{L_1}{2}} (-y_1 y_2 \mathbf{n}_1 + y_1^2 \mathbf{n}_2) dy_1 dy_2 \\
 &= \frac{m}{12} L_1^2 \mathbf{n}_2
 \end{aligned}$$

$$\begin{aligned}
 I_3^{R/R^*} &= \frac{m}{L_1 L_2} \int_{-\frac{L_2}{2}}^{\frac{L_2}{2}} \int_{-\frac{L_1}{2}}^{\frac{L_1}{2}} r \times (n_3 \times r) dy_1 dy_2 = \frac{m}{L_1 L_2} \int_{-\frac{L_2}{2}}^{\frac{L_2}{2}} \int_{-\frac{L_1}{2}}^{\frac{L_1}{2}} (y_1^2 + y_2^2) dy_1 dy_2 \\
 &= \frac{m}{12} (L_1^2 + L_2^2)
 \end{aligned}$$

Hence, we have

$$I^{R/R^*} = \begin{bmatrix} \frac{m}{12} L_2^2 & 0 & 0 \\ 0 & \frac{m}{12} L_1^2 & 0 \\ 0 & 0 & \frac{m}{12} (L_1^2 + L_2^2) \end{bmatrix}$$

(e) Let us compute the inertia matrix of a particle m situated at S^* with respect to O . The vector from S^* to O is given by

$$p = -\frac{L_1}{2} n_1 - \frac{L_2}{2} n_2,$$

so that

$$I_{11}^{S^*/O} = m (p \times n_1) \cdot (p \times n_1) = \frac{m}{4} L_2^2$$

$$I_{12}^{S^*/O} = m (p \times n_1) \cdot (p \times n_2) = -\frac{m}{4} L_1 L_2$$

$$I_{13}^{S^*/O} = m (p \times n_1) \cdot (p \times n_3) = 0$$

$$I_{22}^{s*/o} = m (p \times n_2) \cdot (p \times n_2) = \frac{m}{4} L_1^2$$

$$I_{23}^{s*/o} = m (p \times n_2) \cdot (p \times n_3) = 0$$

$$I_{33}^{s*/o} = m (p \times n_3) \cdot (p \times n_3) = \frac{m}{4} (L_1^2 + L_2^2) .$$

Hence

$$I^{s*/o} = \frac{m}{4} \begin{bmatrix} L_2^2 & -L_1 L_2 & 0 \\ -L_1 L_2 & L_1^2 & 0 \\ 0 & 0 & L_1^2 + L_2^2 \end{bmatrix} ,$$

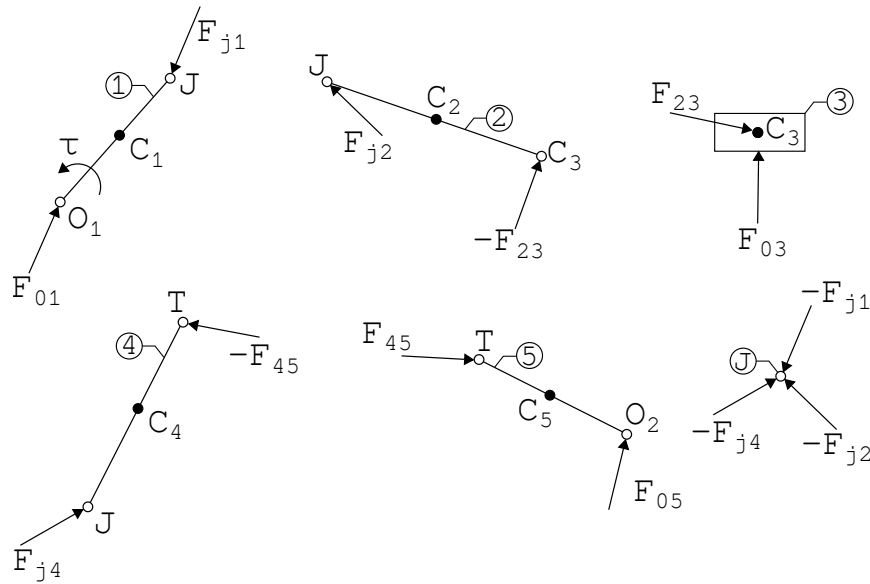
so that

$$I^{R/o} = I^{R/s*} + I^{s*/o}$$

$$= \begin{bmatrix} \frac{m}{12} L_2^2 & 0 & 0 \\ 0 & \frac{m}{12} L_1^2 & 0 \\ 0 & 0 & \frac{m}{12} (L_1^2 + L_2^2) \end{bmatrix} + \frac{m}{4} \begin{bmatrix} L_2^2 & -L_1 L_2 & 0 \\ -L_1 L_2 & L_1^2 & 0 \\ 0 & 0 & L_1^2 + L_2^2 \end{bmatrix}$$

$$= \begin{bmatrix} \frac{m}{3} L_2^2 & -\frac{m}{4} L_1 L_2 & 0 \\ -\frac{m}{4} L_1 L_2 & \frac{m}{3} L_1^2 & 0 \\ 0 & 0 & \frac{m}{3} (L_1^2 + L_2^2) \end{bmatrix} ,$$

which is the same as the result we found in part (d)!



$$F_{01} + F_{j1} - m_1^0 a^{C_1} = 0, \quad (1a)$$

$$F_{j2} - F_{23} - m_2^0 a^{C_2} = 0, \quad (1b)$$

$$F_{23} + F_{03} - m_3^0 a^{C_3} = 0, \quad (1c)$$

$$F_{j4} - F_{45} - m_4^0 a^{C_4} = 0, \quad (1d)$$

$$F_{45} + F_{05} - m_5^0 a^{C_5} = 0, \quad (1e)$$

$$F_{j1} + F_{j2} + F_{j4} = 0, \quad (1f)$$

$$M^{1/C_1} = p^{C_1 O_1} \times F_{01} + p^{C_1 J} \times F_{j1} + \tau - J_1^0 \alpha^1 = 0, \quad (1g)$$

$$M^{2/C_2} = p^{C_2 J} \times F_{j2} - p^{C_2 C_3} \times F_{23} - J_2^0 \alpha^2 = 0, \quad (1h)$$

$$M^{4/C_4} = p^{C_4 J} \times F_{j4} - p^{C_4 T} \times F_{45} - J_4^0 \alpha^4 = 0, \quad (1i)$$

$$M^{5/C_5} = p^{C_5 T} \times F_{45} + p^{C_5 O_2} \times F_{05} - J_5^0 \alpha^5 = 0, \quad (1j)$$

There are 16 scalar equations and 16 unknowns, listed as

$$\xi = (F_{01}, F_{03}, F_{23}, F_{45}, F_{05}, F_{j1}, F_{j2}, F_{j4}, \tau),$$

each of whose x - and y -components are unknown except for F_{03} and τ . The position vectors in equation (1) are given by

$$\begin{aligned}
p^{C_1 O_1} &= -\frac{l_1}{2} \cos(\theta_1) \hat{e}_1 + \sin(\theta_1) \hat{e}_2, & p^{C_1 J} &= \frac{l_1}{2} \cos(\theta_1) \hat{e}_1 + \sin(\theta_1) \hat{e}_2, \\
p^{C_2 J} &= -\frac{l_2}{2} \cos(\theta_2) \hat{e}_1 + \sin(\theta_2) \hat{e}_2, & p^{C_2 C_3} &= \frac{l_2}{2} \cos(\theta_2) \hat{e}_1 + \sin(\theta_2) \hat{e}_2, \\
p^{C_4 J} &= -\frac{l_4}{2} \cos(\theta_4) \hat{e}_1 + \sin(\theta_4) \hat{e}_2, & p^{C_4 T} &= \frac{l_4}{2} \cos(\theta_4) \hat{e}_1 + \sin(\theta_4) \hat{e}_2, \\
p^{C_5 T} &= -\frac{l_5}{2} \cos(\theta_5) \hat{e}_1 + \sin(\theta_5) \hat{e}_2, & p^{C_5 O_2} &= \frac{l_5}{2} \cos(\theta_5) \hat{e}_1 + \sin(\theta_5) \hat{e}_2.
\end{aligned}$$

Refer to Figure 2 for questions 2, 3, and 4. The link lengths are given by $l_0 = 0.25$ m, $l_1 = l_4 = 0.3$ m, $l_2 = l_3 = 0.35$ m. Each link is made of homogeneous material, so assume that the center of mass of each link is at its geometric center. The inertial constants are given by $m_1 = m_4 = 0.15$ kg, $m_2 = m_3 = 0.1$ kg and $I_1 = I_4 = 1.5 \times 10^{-3}$ kg $\cdot$ m², $I_2 = I_3 = 2 \times 10^{-3}$ kg $\cdot$ m². Gravity acts downwards within the plane.

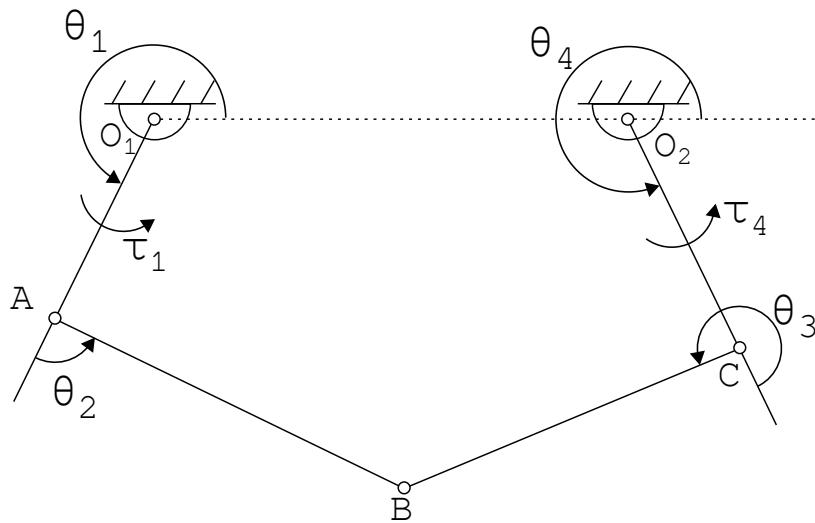
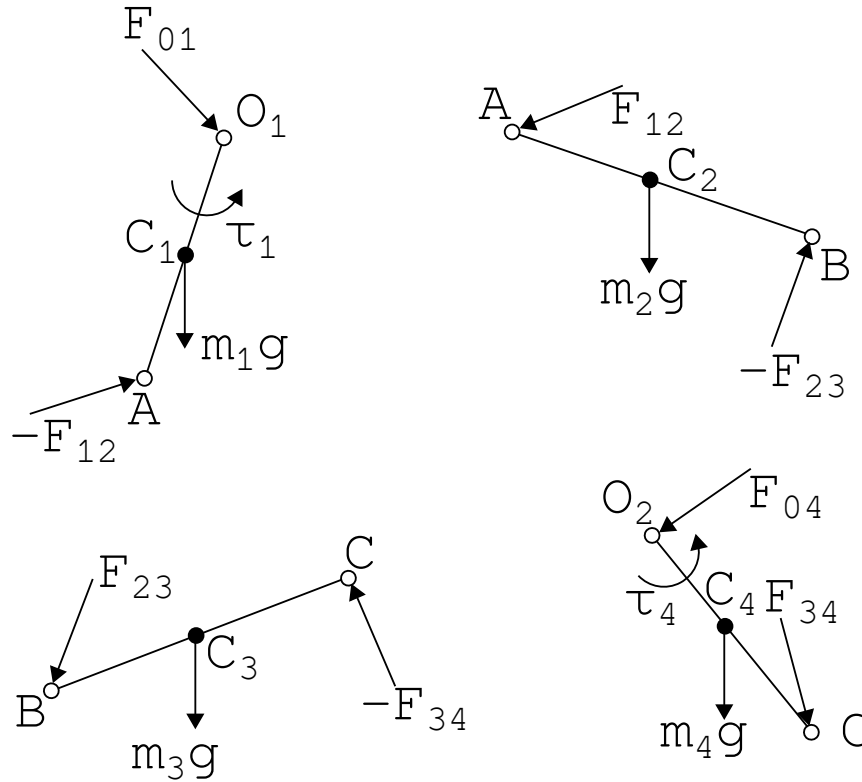


Figure 2: Pantograph

2. [25 points] Perform position-level kinematic analysis for the pantograph depicted in Figure 2. Follow this analysis up by performing a static-force analysis. Find the values of the motor torque $\tau = [\tau_1 \ \tau_4]^T$ necessary to “cancel the gravity” (i.e. keep the mechanism stationary) when $[\theta_1 \ \theta_4]^T = [240^\circ \ 310^\circ]^T$. (Notice that you first need to find what θ_2 and θ_3 correspond to these values of θ_1 and θ_4 .)

Solution: Solving the position-level forward kinematics with a numerical technique yields $\theta_2 = 90.890^\circ$ and $\theta_3 = -95.098^\circ$.

The figure below shows the free body diagrams of each link, excluding the inertial forces.



The Newton-Euler formulation for static equilibrium states that the sum of all the forces and the sum of the moments about any point on a rigid body must equal zero.

The sum of the forces R_i on rigid body i is given by

$$R_i = F_{i-1,i} - F_{i,i+1} - m_i g \hat{e}_2 = 0, \quad i \in \{1, 2, 3\}, \quad (2a)$$

$$R_4 = F_{34} + F_{04} - m_4 g \hat{e}_2 = 0. \quad (2b)$$

The sum of the moments about the center of mass of each link are given by

$$M^{1/C_1} = p^{C_1 O_1} \times F_{01} - p^{C_1 A} \times F_{12} + \tau_1 e_3 = 0, \quad (3a)$$

$$M^{2/C_2} = p^{C_2 A} \times F_{12} - p^{C_2 B} \times F_{23} = 0, \quad (3b)$$

$$M^{3/C_3} = p^{C_3 B} \times F_{23} - p^{C_3 C} \times F_{34} = 0, \quad (3c)$$

$$M^{4/C_4} = p^{C_4 C} \times F_{34} + p^{C_4 O_2} \times F_{04} + \tau_4 e_3 = 0, \quad (3d)$$

where the various position vectors are given in coordinates by

$$\begin{aligned} p^{C_1O_1} &= -p^{C_1A} = -\frac{l_1}{2} [\cos(\theta_1) \quad \sin(\theta_1) \quad 0]^\top, \\ p^{C_2A} &= -p^{C_2B} = -\frac{l_2}{2} [\cos(\theta_1 + \theta_2) \quad \sin(\theta_1 + \theta_2) \quad 0]^\top, \\ p^{C_3B} &= -p^{C_3C} = \frac{l_3}{2} [\cos(\theta_3 + \theta_4) \quad \sin(\theta_3 + \theta_4) \quad 0]^\top, \\ p^{C_4C} &= -p^{C_4O_2} = \frac{l_4}{2} [\cos(\theta_4) \quad \sin(\theta_4) \quad 0]^\top. \end{aligned}$$

Putting equations (2) and (3) into matrix form yields the system $Ax = b$ (3 significant digits are displayed) where $x = [F_{01} \quad F_{12} \quad F_{23} \quad F_{34} \quad F_{04} \quad \tau]^\top$ and

$$A = \begin{bmatrix} 1 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & -1 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 & -1 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 & 0 & -1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 & 1 & 0 & 0 \\ -0.13 & 0.075 & -0.13 & 0.075 & 0 & 0 & 0 & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & -0.085 & -0.153 & -0.085 & -0.153 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0.1 & -0.144 & 0.1 & -0.144 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0.115 & 0.096 & -0.115 & -0.096 & 0 & 1 \end{bmatrix},$$

$$b = [0 \quad 1.472 \quad 0 \quad 0.981 \quad 0.0 \quad 0.981 \quad 0 \quad 1.472 \quad 0 \quad 0 \quad 0 \quad 0]^\top.$$

Solving this system of linear equations yields, in particular,

$$\tau = \begin{bmatrix} -0.45242 \\ 0.52138 \end{bmatrix}.$$

3. [25 points] Complete the kinematic analysis of the pantograph by performing a velocity- and acceleration-level analysis. Given that $[\theta_1 \quad \theta_4]^\top = [240^\circ \quad 310^\circ]^\top$ and $[\dot{\theta}_1 \quad \dot{\theta}_4]^\top = [-15 \quad 5]^\top \frac{\text{rad}}{\text{s}}$, at an instant in time, find $[\theta_2 \quad \theta_3]^\top$ and $[\dot{\theta}_2 \quad \dot{\theta}_3]^\top$. Additionally, find the angular acceleration of all links at this instant if $\tau = [\tau_1 \quad \tau_4]^\top = [1 \quad -1]^\top \text{ Nm}$.

Solution: Solving the velocity-level kinematics with the given velocities of links 1 and 4 yields

$$\begin{bmatrix} \dot{\theta}_2 \\ \dot{\theta}_3 \end{bmatrix} = \begin{bmatrix} 25.816 \\ -21.002 \end{bmatrix} \frac{\text{rad}}{\text{s}}.$$

We append the static-force balance equations the inertial forces and change our unknown from τ to $\ddot{\theta} = [\ddot{\theta}_1 \ \ddot{\theta}_2 \ \ddot{\theta}_3 \ \ddot{\theta}_4]^\top$. Thus equations (2) and (3) become

$$R_i = F_{i-1,i} - F_{i,i+1} - m_i g \hat{e}_2 - m_i {}^0a^{C_i} = 0, \quad i \in \{1, 2, 3\}, \quad (4a)$$

$$R_4 = F_{34} + F_{04} - m_4 g \hat{e}_2 - m_4 {}^0a^{C_4} = 0, \quad (4b)$$

$$M^{1/C_1} = p^{C_1 O_1} \times F_{01} - p^{C_1 A} \times F_{12} + \tau_1 e_3 - I_1 {}^0\alpha^1 = 0, \quad (4c)$$

$$M^{2/C_2} = p^{C_2 A} \times F_{12} - p^{C_2 B} \times F_{23} - I_2 {}^0\alpha^2 = 0, \quad (4d)$$

$$M^{3/C_3} = p^{C_3 B} \times F_{23} - p^{C_3 C} \times F_{34} - I_3 {}^0\alpha^3 = 0, \quad (4e)$$

$$M^{4/C_4} = p^{C_4 C} \times F_{34} + p^{C_4 O_2} \times F_{04} + \tau_4 e_3 - I_4 {}^0\alpha^4 = 0, \quad (4f)$$

where $\alpha_1 = \ddot{\theta}_1 \hat{e}_3$, $\alpha_2 = (\ddot{\theta}_1 + \ddot{\theta}_2) \hat{e}_3$, $\alpha_3 = (\ddot{\theta}_3 + \ddot{\theta}_4) \hat{e}_3$, and $\alpha_4 = \ddot{\theta}_4 \hat{e}_3$. In these equations, the accelerations of the centers of masses are given by

$$\begin{aligned} {}^0a^{C_1} &= {}^0\alpha^1 \times p^{O_1 C_1} + {}^0\omega^1 \times ({}^0\omega^1 \times p^{O_1 C_1}), \\ {}^0a^{C_2} &= {}^0\alpha^1 \times p_1 + {}^0\omega^1 \times ({}^0\omega^1 \times p_1) + {}^0\alpha^2 \times p^{AC_2} + {}^0\omega^2 \times ({}^0\omega^2 \times p^{AC_2}), \\ {}^0a^{C_3} &= -{}^0\alpha^4 \times p_4 - {}^0\omega^4 \times ({}^0\omega^4 \times p_4) + {}^0\alpha^3 \times p^{CC_3} + {}^0\omega^3 \times ({}^0\omega^3 \times p^{CC_3}), \\ {}^0a^{C_4} &= {}^0\alpha^4 \times p^{O_2 C_4} + {}^0\omega^4 \times ({}^0\omega^4 \times p^{O_2 C_4}), \end{aligned}$$

where $p_1 = \overrightarrow{O_1 A}$, $p_2 = \overrightarrow{AB}$, $p_3 = \overrightarrow{BC}$, $p_4 = \overrightarrow{CO_2}$. We need two more equations to be able to solve for all the unknowns. These come from the acceleration-level kinematic analysis as

$$\sum_{k=1}^4 {}^0\alpha^k \times p_k + {}^0\omega^k \times ({}^0\omega^k \times p_k) = 0. \quad (5)$$

After appending equations (4) and (5), we put them into matrix form $A_{dyn}x = b_{dyn}$ and solve for the unknowns $x = [F_{01} \ F_{12} \ F_{23} \ F_{34} \ F_{04} \ \ddot{\theta}]$. Let us decompose $A_{dyn} \in \mathbb{R}^{14 \times 14}$ as follows

$$A_{dyn} = \begin{bmatrix} A_1 & A_2 \\ 0 & A_3 \end{bmatrix},$$

where $A_1 \in \mathbb{R}^{12 \times 10}$, $A_2 \in \mathbb{R}^{12 \times 4}$, $0 \in \mathbb{R}^{2 \times 10}$, and $A_3 \in \mathbb{R}^{2 \times 4}$. Here, $A_1 = A \begin{bmatrix} I_{10} \\ 0 \end{bmatrix}$ where I and 0 are the identity and zero matrices of appropriate sizes and A is taken from problem 2. The remaining matrices A_2 and A_3 are given by (up to 4 digits)

$$A_2 = \begin{bmatrix} -0.0195 & 0 & 0 & 0 \\ 0.0112 & 0 & 0 & 0 \\ -0.0345 & -0.0085 & 0 & 0 \\ -0.0003 & -0.0153 & 0 & 0 \\ 0 & 0 & -0.01 & -0.033 \\ 0 & 0 & 0.0144 & -0.0049 \\ 0 & 0 & 0 & -0.0172 \\ 0 & 0 & 0 & -0.0145 \\ -0.0015 & 0 & 0 & 0 \\ -0.002 & -0.002 & 0 & 0 \\ 0 & 0 & -0.002 & -0.002 \\ 0 & 0 & 0 & -0.0015 \\ 0.4301 & 0.1703 & -0.2003 & -0.4301 \\ 0.1558 & 0.3058 & 0.287 & 0.0942 \end{bmatrix}, \quad A_3 = \begin{bmatrix} 0.43 & 0.17 & -0.2 & -0.43 \\ 0.156 & 0.306 & 0.287 & 0.094 \end{bmatrix}.$$

The vector $b_{dyn} \in \mathbb{R}^{14}$ turns out to be (up to 3 digits)

$$b_{dyn} = [2.531 \quad 5.856 \quad 1.586 \quad 7.823 \quad 3.193 \quad 4.119 \quad -0.3616 \quad 1.902 \quad -1 \quad 0 \quad 0 \quad 1 \quad 70.702 \quad -21.351]^\top$$

Solving $A_{dyn}x = b_{dyn}$ and recording the last 4 elements, which correspond to the second time derivatives of the joint angles, we find

$$[\ddot{\theta}_1 \quad \ddot{\theta}_2 \quad \ddot{\theta}_3 \quad \ddot{\theta}_4]^\top = [160.843 \quad -304.665 \quad 240.37 \quad -236.095]^\top \frac{\text{rad}}{\text{s}^2}.$$

4. [25 points (bonus)] (Strongly recommended as this will help with your project immensely)

Assume that the motor torques equal identically to $\tau = [-0.45 \quad 0.45]^\top$. Suppose at the initial time ($t = 0$) the mechanism's state is $[\theta_1 \quad \theta_4 \quad \dot{\theta}_1 \quad \dot{\theta}_4] = [240^\circ \quad 310^\circ \quad 0 \quad 0]$. Write a Julia or Matlab program that simulates the system for 20 seconds (finds θ_1 , θ_4 , $\dot{\theta}_1$, and $\dot{\theta}_4$ for all time $t \in [0, 20]$ using Newton-Euler equations). You may use the Julia package `DifferentialEquations`, Matlab's `ode45` function or simply implement a backward Euler integration scheme for numerical integration. Present the simulation results by plotting θ_1 , θ_4 , $\dot{\theta}_1$, and $\dot{\theta}_4$ as a function of time t .

Solution: The simulation results in the following evolution of system states

